

Control theory: state-space approach for linear systems

Discrete-time systems

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Linear systems described by state-space representations

$$\begin{cases} x(k+1) = Ax(k) + Bu(k) \\ y(k) = Cx(k) \end{cases} \quad (1)$$

in which the time, denoted k , takes some discrete values (e.g. \mathbb{Z}).

From a TF to a state-space repr.

Same observation as in the continuous-time case :

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Same observation as in the continuous-time case :

- ▶ A given system admits several equivalent state-space representations.
- ▶ Consequently, from a transfer function there exist several methods to get the distinct but equivalent state-space representations of a system, each of them having a particular form.

From a TF to a state-space repr.

Controllable Canonical Form

$$\left\{ \begin{array}{l} x(k+1) = \begin{pmatrix} 0 & 1 & 0 & \dots & 0 \\ 0 & 0 & 1 & \dots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & \dots & 1 \\ -a_0 & -a_1 & -a_2 & \dots & -a_{n-1} \end{pmatrix} x(k) + \begin{pmatrix} 0 \\ 0 \\ \vdots \\ 0 \\ 1 \end{pmatrix} u(k) \\ y(k) = (b_0 \ b_1 \ \dots \ b_{n-1}) x(k) \end{array} \right. .$$

From a TF to a state-space repr.

Observable Canonical Form

$$\left\{ \begin{array}{l} x(k+1) = \begin{pmatrix} -a_{n-1} & 1 & 0 & \dots & 0 \\ -a_{n-2} & 0 & 1 & \dots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ -a_1 & 0 & 0 & \dots & 1 \\ -a_0 & 0 & 0 & \dots & 0 \end{pmatrix} x(k) + \begin{pmatrix} b_{n-1} \\ b_{n-2} \\ \vdots \\ b_1 \\ b_n \end{pmatrix} u(k) \\ y(k) = (1 \ 0 \ \dots \ 0) x(k) \end{array} \right. .$$

Solution to state-space equations

Theorem

$x(k_0)$ value of the state at initial time-instant k_0 , state at k is given by

$$x(k) = A^{k-k_0}x(k_0) + \sum_{l=k_0}^{k-1} A^{k-1-l}Bu(l), \quad (2)$$

and the output can be written

$$y(k) = CA^{k-k_0}x(k_0) + \sum_{l=k_0}^{k-1} CA^{k-1-l}Bu(l). \quad (3)$$

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The proof proceeds by induction.

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Definition

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If $u(k) = 0$ for $k > k_1$, state for $k \geq k_1$ can be written :

$$\begin{aligned}x(k) &= A^{k-k_1}x(k_1) + \sum_{l=k_1}^{k-1} A^{k-1-l}Bu(l) \\ &= A^{k-k_1}x(k_1) \quad (\text{since } u(l) = 0 \text{ for } l \geq k_1),\end{aligned}$$

which tends to zero while $k \rightarrow \infty$ if $\lim_{k \rightarrow \infty} A^{k-k_1} = \mathbf{0}$.

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Theorem

A linear system is said to be stable if, and only if, all the eigenvalues of its matrix evolution have a modulus strictly less than 1.

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The *characteristic polynomial* $P(z)$ of a linear system is defined by

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Corollary (Stability criterion)

A linear system is said to be stable if, and only if, all the roots of its characteristic polynomial have a modulus strictly less than 1.

Coontrollability

Analogous to the time-continuous case!

Theorem (Controllability criterion)

A linear system is said to be controllable if, and only if,

$$\text{rank} \left(\overbrace{B|AB|A^2B|\dots|A^{n-1}B}^{\Gamma_{com}} \right) = n, \quad (5)$$

where n is the dimension of matrix evolution A .

Observability

Analogous to the time-continuous case!

Theorem (Observability criterion)

A linear system is said to be observable if, and only if,

$$\text{rank} \begin{pmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{pmatrix} = n. \quad (6)$$

Continuous-time systems studied by means of a computer : sampled discrete-time systems

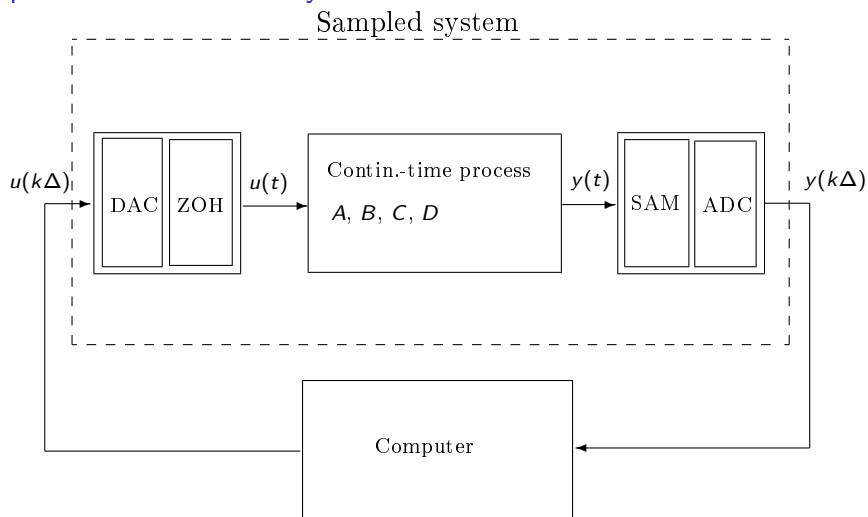


Figure – Continuous-time system seen as a sampled discrete-time system by the

Sampled systems : discretizing a continuous-time state-equation

- We consider a continuous-time state-space representation :

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- When it is sampled, it can be described at sampling time-instants $k\Delta$, denoted k to lighten equations, by discrete-time equations :

$$\begin{cases} x(k+1) &= A_{ech}x(k) + B_{ech}u(k) \\ y(k) &= C_{ech}x(k) \end{cases} \quad (8)$$

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- We want to figure out matrices A_{ech} , B_{ech} and C_{ech} from known matrices A , B and C .

Sampled systems : discretizing a continuous-time state-equation

Knowing the state-value at sampling-time $t_k = k\Delta$, makes it possible to find state at next sampling-time $t_{k+1} = (k + 1)\Delta$:

$$x(t_{k+1}) = e^{A(t_{k+1}-t_k)}x(t_k) + \int_{t_k}^{t_{k+1}} e^{A(t_{k+1}-\tau)}Bu(\tau)d\tau . \quad (9)$$

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The ZOH keeps $u(t)$ at a constant value $u(t_k) = u(k\Delta)$, denoted $u(k)$, during time-interval $[t_k, t_{k+1}[$, we then have :

$$u(\tau) = u(k) \text{ pour } \tau \in [t_k, t_{k+1}[.$$

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Let us state $\nu = \tau - t_k$, and since $\Delta = t_{k+1} - t_k$, (9) leads to :

$$x(k + 1) = e^{A\Delta}x(k) + \int_0^{\Delta} e^{A(\Delta-\nu)}d\nu Bu(k) . \quad (10)$$

Sampled systems : discretizing a continuous-time state-equation

By identifying terms, we obtain :

$$A_{ech} = e^{A\Delta} \quad (11)$$

$$B_{ech} = \int_0^{\Delta} e^{A(\Delta-\nu)} d\nu B \quad (12)$$

$$C_{ech} = C \quad (13)$$

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$$C_{ech} = C \quad (13)$$

- Matrices A_{ech} and B_{ech} depend on sampling period Δ , and have to be evaluated as soon as the sampling period changes.

Sampled systems : choice of the sampling period

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- if Δ is too large, important pieces of information are lacking on the evolution of the system from the computer point of view.
- if Δ is too small, the computer is excessively "loaded". If the system does evolve only a little bit between two successive samples, the piece of information obtained by a new sample does not bring much and the processor is then requested pointlessly.

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The choice of sampling period Δ depends on the dynamics of the system.

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Shannon's

$$f_e \geq 2f_h$$

in which f_e is the sampling frequency and f_h is the highest frequency to be kept into the signal.

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Practical rule applied in control theory

$$5f_h < f_e < 25f_h$$

On one side, it is an "over-sampling" compared to Shannon's theorem. On the other side, this rule gives a bound to avoid to request too often and unnecessarily the processor.

Sampled systems : choice of the sampling period

Consequently, we select

$$5f_h < f_e < 25f_h .$$

	First-order system	Second-order system
With	$f_h \approx \frac{1}{2\pi T}$	$f_h \approx \frac{\omega_n}{2\pi}$
We then get	$0.25 T < \Delta < 1.25 T$	$0.25 < \Delta \omega_n < 1.25$

Example : container handling gantry crane

State-space representation and transfer function :

$$\begin{cases} \dot{x}(t) &= \begin{pmatrix} 0 & g \\ -\frac{1}{L} & 0 \end{pmatrix} x(t) + \begin{pmatrix} 0 \\ \frac{1}{L} \end{pmatrix} u(t) \\ y(t) &= \begin{pmatrix} 1 & 0 \end{pmatrix} x(t) \end{cases}$$

$$H(s) = C(sI - A)^{-1}B = \frac{1}{1 + \frac{L}{g}s^2}.$$

It is a second-order system for which the natural frequency is given by $\omega_n = \sqrt{\frac{g}{L}}$.
Sampling period Δ can then be chosen into interval

$$\left] 0.25\sqrt{\frac{L}{g}}, 1.25\sqrt{\frac{L}{g}} \right[$$

Example : container handling gantry crane

State-space representation of the system sampled at period Δ :

Using formula $e^{At} = \mathcal{L}^{-1} [(sI - A)^{-1}]$, we have¹ :

$$\begin{aligned} e^{At} &= \mathcal{L}^{-1} \left[\begin{pmatrix} \frac{s}{s^2 + g/L} & \frac{g}{s^2 + g/L} \\ \frac{1/L}{s^2 + g/L} & \frac{s}{s^2 + g/L} \end{pmatrix} \right] \\ &= \begin{pmatrix} \cos(\sqrt{\frac{g}{L}}t) & \sqrt{gL} \sin(\sqrt{\frac{g}{L}}t) \\ -\frac{1}{\sqrt{gL}} \sin(\sqrt{\frac{g}{L}}t) & \cos(\sqrt{\frac{g}{L}}t) \end{pmatrix} \end{aligned}$$

Hence

$$A_{ech} = \begin{pmatrix} \cos(\sqrt{\frac{g}{L}}\Delta) & \sqrt{gL} \sin(\sqrt{\frac{g}{L}}\Delta) \\ -\frac{1}{\sqrt{gL}} \sin(\sqrt{\frac{g}{L}}\Delta) & \cos(\sqrt{\frac{g}{L}}\Delta) \end{pmatrix}$$

1. As a reminder, Laplace transform of :

- $\sin(\omega t)$ is $\frac{\omega}{s^2 + \omega^2}$;
- $\cos(\omega t)$ is $\frac{s}{s^2 + \omega^2}$.

Example : container handling gantry crane

We get $^2 B_{ech} = \int_0^\Delta e^{A(\Delta-\nu)} d\nu B :$

$$\begin{aligned} B_{ech} &= \int_0^\Delta \begin{pmatrix} \cos(\sqrt{\frac{g}{L}}\Delta - \sqrt{\frac{g}{L}}\nu) & \sqrt{gL} \sin(\sqrt{\frac{g}{L}}\Delta - \sqrt{\frac{g}{L}}\nu) \\ -1/\sqrt{gL} \sin(\sqrt{\frac{g}{L}}\Delta - \sqrt{\frac{g}{L}}\nu) & \cos(\sqrt{\frac{g}{L}}\Delta - \sqrt{\frac{g}{L}}\nu) \end{pmatrix} d\nu B \\ &= \left[\begin{pmatrix} -\sqrt{\frac{g}{L}} \sin(\sqrt{\frac{g}{L}}\Delta - \sqrt{\frac{g}{L}}\nu) & -L \cos(\sqrt{\frac{g}{L}}\Delta - \sqrt{\frac{g}{L}}\nu) \\ \frac{1}{g} \cos(\sqrt{\frac{g}{L}}\Delta - \sqrt{\frac{g}{L}}\nu) & -\sqrt{\frac{g}{L}} \sin(\sqrt{\frac{g}{L}}\Delta - \sqrt{\frac{g}{L}}\nu) \end{pmatrix} \right]_0^\Delta B \\ &= \begin{pmatrix} -1 + \cos(\sqrt{\frac{g}{L}}\Delta) \\ \frac{1}{\sqrt{gL}} \sin(\sqrt{\frac{g}{L}}\Delta) \end{pmatrix} \end{aligned}$$

$$C_{ech} = C = \begin{pmatrix} 1 & 0 \end{pmatrix}$$

2. As a reminder, the derivative of :

- $\frac{1}{\omega} \sin(\omega t + \phi)$ is $\cos(\omega t + \phi)$;
- $-\frac{1}{\omega} \cos(\omega t + \phi)$ is $\sin(\omega t + \phi)$.